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### **TURBOMACHINERY AERODYNAMICS In this lecture...**

- Computational Fluid Dynamics for turbomachinery
	- Introduction and overview
	- Grid generation
	- Boundary conditions

- Computational Fluid Dynamics (CFD) is a powerful analytical tool.
- Is a third approach for analysis besides experimental approach and theoretical approach.
- CFD compliments theory and experiments and is not primarily intended to replace these.
- CFD is currently a commonly used research tool.
- CFD is an essential component of the design, analysis and optimization cycle.

- There are various levels of CFD analysis
	- Simple Euler (potential flow) solutions
	- 2-D/axisymmetric Navier-Stokes solution
	- 3-D Navier-Stokes solution
		- Reynolds Averaged Navier-Stokes (RANS) and Unsteady RANS (URANS)
		- Large Eddy Simulation (LES)
		- Direct Numerical Simulation (DNS)
- CFD analysis could also be
	- Steady or unsteady
	- Incompressible or compressible
	- Laminar or turbulent
	- Internal or external flow

- CFD involves solving the fundamental governing equations of fluid flow:
	- Conservation of mass
	- Conservation of momentum
	- Conservation of energy
	- Equation of state
	- Species conservation (reacting flows)

- Steps in CFD solution
	- Setting up the domain
	- Discretisation of the domain in space and time (for unsteady solution)
	- Defining boundary conditions
	- Solving the appropriate governing equations for the domain on the discretised points
	- Post-processing and analysis of the converged solution.

- Turbomachinery: complex shear flows
	- Shear layers on rotating surfaces
	- Shear layers developing on curved surfaces
	- Separated flows: shock-boundary layer interaction, corner separation…
	- Swirling flows and vortices
	- Interacting boundary layers

- Challenges in turbomachinery CFD
	- Grid generation
		- Complex geometry
		- Rotating domain
	- Flow is 3-D, highly unsteady, rotating, and turbulent
		- Capturing the losses and other viscous effects
		- Turbulence modelling
	- Fluid-structure interactions

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- Types of simulations
	- 2D, quasi-3D, 3D
	- $\cdot$  2D
		- Conceptual design phase
		- Long blades/vanes (LP turbines)
		- Reasonable results
	- Quasi-3D
		- Area of flow path changes
		- Extra source terms for acceleration/deceleration or boundary layer growth

- Types of simulations
	- 3D
		- True geometry required
		- Simulate secondary flows, shock locations
		- End wall boundary layers
	- Transient or stationery
		- Stationery simulations more common
		- Transient: flow unsteadiness, vortex shedding, wake interaction with rotors

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- Solver
	- Euler
	- 3D NS
	- RANS, URANS
	- DES, DDES
	- •
	- •
	- LES
	- •
	- DNS

- Grid/mesh
	- Structured, unstructured and hybrid grids
- Structured grid
	- More suited for well-defined geometries
	- More difficult to generate
	- Easier to control near-wall clustering of cells
- Unstructured grid
	- Primarily intended for complex geometries
	- Easier to generate
	- Not much control over the near-wall clustering of cells
	- Easily automated

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### **Grid Generation**



### Structured grid with multiple blocks

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### **Grid Generation**



#### Unstructured grid

- Multi-block structured grid
	- In order to generate structured grid over curved surfaces, multiple blocks need to be defined.
	- Interface of the blocks need to be carefully managed.
	- Grid topology needs to be appropriately defined.
	- The Grid topology also needs to account for the change in geometry of the blade from hub to tip.

- Topology
	- Is a structure off blocks that acts as a framework for placing mesh elements.
	- Blocks are laid out without gaps with shared edges and corners.
	- Blocks contain same number of elements along each side.
	- Is usually invariant from hub to tip.
	- Can be edited on 2-D layers from hub to tip sections.
	- Number of blocks will dictate the skewness of the grid elements.

- Grid topology schemes
	- O-grid:
		- Usually used around the blade by forming a continuous loop around it
		- Yields excellent boundary layer resolution
		- gives good control over the y+ values that needs to be tightly monitored
		- Provides near orthogonal elements on the blades

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### **Grid Generation**



#### O-grid topology

- J-grid:
	- Usually used near leading and trailing edges
	- Wraps up in opposite directions at the leading and trailing edges
- H-grid:
	- Tends to complete the meshing by adding some blocks in an unstructured manner
	- The structured blocks extend from upstream of the LE, downstream of the TE and between the blades and the periodic surfaces



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### **Grid Generation**





#### J-grid topology H-grid topology

- Other topology options include C-grid and L-grid.
- These are also often used at the leading and trailing edges.
- All the above grid topologies are used along with an O-grid for proper resolution of the boundary layer.
- Proper resolution of the leading and trailing edge radii are important.
- Establishing grid-independence or grid-insensitivity of the results is now a standard practice.

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### **Grid Generation**





#### L-grid topology C-grid topology

- To capture the flow physics correctly, the boundary conditions must be set appropriately.
- Quality of the solutions is a strong function of the boundary conditions.
- Turbomachinery flows
	- Inlet boundary
	- Exit boundary
	- Periodic boundary
	- Walls or surfaces

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### **Boundary conditions**



#### Typical flow domain with the boundaries

- Inlet boundary conditions
	- Depends upon the application
	- Flow conditions (incompressible or compressible)
	- Total pressure, total temperature, velocity components/profile (most commonly used)
	- There are other forms of specifying the inlet boundary conditions: velocity inlet, mass flow inlet etc.: not commonly used due to several limitations.

- Exit boundary conditions
	- Exit static pressure to achieve the required mass flow
	- It is also possible to specify a static pressure distribution at the exit domain.
	- Alternatively, mass flow can be directly specified at the exit.
	- For incompressible flows, using either of the two does not affect the results.
	- However, for compressible flows, static pressure outlet condition yields better results.

- For single passage simulations, periodic boundary conditions are used for simulating the effect of a blade row.
- The domain must be appropriately chosen to ensure that periodic boundary conditions are indeed valid.
- On surfaces (blade, hub and shroud), no-slip and adiabatic conditions are usually used.
- In turbines with hot gases present, the adiabatic condition may be replaced by constant heat flux condition.

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### **Boundary conditions**



#### Typical flow domain with the boundaries

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**TURBOMACHINERY AERODYNAMICS**