Lecture – 28

Linear Quadratic Regulator (LQR) Design – II

Dr. Radhakant Padhi

Asst. Professor

Dept. of Aerospace Engineering Indian Institute of Science - Bangalore



Outline

- Stability and Robustness properties of LQR
- Optimum value of the cost function
- Extension of LQR design
 - For cross-product term in cost function
 - Rate of state minimization
 - Rate of control minimization
 - LQR design with prescribed degree of stability
- LQR for command tracking
- LQR for inhomogeneous systems

Stability and Robustness Properties

Dr. Radhakant Padhi

Asst. Professor

Dept. of Aerospace Engineering

Indian Institute of Science - Bangalore



LQR Design: Stability of Closed Loop System

- Closed loop system $\dot{X} = AX + BU = (A BK)X$
- Lyapunov function $V(X) = X^T P X$

$$\dot{V} = \dot{X}^T P X + X^T P \dot{X}
= \left[\left(A - B K \right) X \right]^T P X + X^T P \left[\left(A - B K \right) X \right]
= X^T \left[\left(A - B R^{-1} B^T P \right)^T P + P \left(A - B R^{-1} B^T P \right) \right] X
= X^T \left[\left(P A + A^T P - P B R^{-1} B^T P + Q \right) - Q - P B R^{-1} B^T P \right] X
= X^T \left[-Q - P B R^{-1} B^T P \right] X$$

LQR Design: Stability of Closed Loop System

For
$$R > 0$$
, $R^{-1} > 0$. Also $P > 0$

So
$$PBR^{-1}B^TP > 0$$

Also
$$Q \ge 0$$
.

Hence,
$$(PBR^{-1}B^TP + Q) > 0$$

$$\dot{V}(X) < 0$$

Hence, the closed loop system is always asymptotically stable!

LQR Design: Minimum value of cost function

$$J = \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q X + U^T R U \right) dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + \left(-R^{-1} B^T P X \right)^T R \left(-R^{-1} B^T P X \right) \right] dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} X^T \left(Q + P B R^{-1} B^T P \right) X dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left(-\dot{V} \right) dt = -\frac{1}{2} \left[V \right]_{t_0}^{\infty} = -\frac{1}{2} \left[X^T P X \right]_{t_0}^{\infty}$$

$$= \frac{1}{2} \left[X_0^T P X_0 - X_{\infty}^T P X_{\infty} \right] = \frac{1}{2} \left(X_0^T P X_0 \right)$$

LQR Design: Robustness of Closed Loop System

■ Gain Margin: ∞

Phase Margin: 60°

(Ref.: D. S. Naidu, Optimal Control Systems, CRC Press, 2003.)

Extensions of LQR Design

Dr. Radhakant Padhi

Asst. Professor

Dept. of Aerospace Engineering

Indian Institute of Science - Bangalore



1. Cross Product Term in P.I.

$$J = \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q X + 2 X^T W U + U^T R U \right) dt$$

Let us consider the expression:

$$X^{T} (Q - WR^{-1}W^{T}) X + (U + R^{-1}W^{T}X)^{T} R(U + R^{-1}W^{T}X)$$

$$= X^{T}QX + U^{T}RU + (U^{T}W^{T}X + X^{T}WU)$$

$$= X^{T}QX + 2X^{T}WU + U^{T}RU$$

LQR Extensions: 1. Cross Product Term in P.I.

$$J = \frac{1}{2} \int_{t_0}^{\infty} \left[X^T \left(Q - WR^{-1}W^T \right) X + \left(U + R^{-1}W^T X \right)^T R \left(U + R^{-1}W^T X \right) \right] dt$$

$$=\frac{1}{2}\int_{t_0}^{\infty} \left(X^T Q_1 X + U_1 R U_1\right) dt$$

$$\dot{X} = AX + BU$$

$$= AX + B\left(U_{1} - R^{-1}W^{T}X\right)$$

$$= \left(A - BR^{-1}W^{T}\right)X + BU_{1}$$

$$= A_{1}X + BU_{1}$$

Control Solution

$$U_1 = -KX$$

$$U = U_1 - R^{-1}W^T X$$
$$= -\left(K + R^{-1}W^T\right) X$$

2. Weightage on Rate of State

$$J = \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q X + U^T R U + \dot{X}^T S \dot{X} \right) dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + U^T R U + \left(A X + B U \right)^T S \left(A X + B U \right) \right] dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + U^T R U + X^T A^T S A X + X^T A^T S B U \right] dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + U^T R U + X^T A^T S B U \right] dt$$

$$= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + U^T A X + U$$

3. Weightage on Rate of Control

$$J = \frac{1}{2} \int_{0}^{\infty} \left(X^{T} Q X + U^{T} R U + \dot{U}^{T} \hat{R} \dot{U} \right) dt$$

Let
$$\mathbf{X} = \left\lceil \frac{X}{U} \right\rceil$$
, $V = \dot{U}$

$$J = \frac{1}{2} \int_{0}^{\infty} \left(\mathbf{X}^{T} \begin{bmatrix} Q & 0 \\ 0 & R \end{bmatrix} \mathbf{X} + V^{T} \hat{R} V \right) dt$$

$$J = \frac{1}{2} \int_{0}^{\infty} \left(\mathbf{X}^{T} \hat{Q} \mathbf{X} + V^{T} \hat{R} V \right) dt$$

3. Weightage on Rate of Control

$$\dot{X} = AX + BU, \quad X(0) = X_0$$

$$\dot{U} = V$$

$$\dot{X} = \begin{bmatrix} A & B \\ 0 & 0 \end{bmatrix} \mathbf{X} + \begin{bmatrix} 0 \\ I \end{bmatrix} V = \hat{A}\mathbf{X} + \hat{B}V$$

Note:

- (1) The dimension of the problem has increased from n to (n+m)
- (2) If $\{A, B\}$ is controllable, it can be shown that the new system is also controllable.

LQR Extensions: 3. Weightage on Rate of Control

Solution:

$$V = \dot{U} = -\hat{R}^{-1}\hat{B}^T\hat{P}\mathbf{X}$$

where \hat{P} is the solution of

$$\hat{A}^{T}\hat{P} + \hat{P}A - \hat{P}\hat{B}\hat{R}^{-1}\hat{B}^{T}\hat{P} + \hat{Q} = 0$$

Hence

$$\dot{U} = -\hat{R}^{-1} \begin{bmatrix} 0 & I \end{bmatrix} \begin{bmatrix} \hat{P}_{11} & \hat{P}_{12} \\ \hat{P}_{12}^T & \hat{P}_{22} \end{bmatrix} \mathbf{X} = -\hat{R}^{-1} \begin{bmatrix} \hat{P}_{12}^T & \hat{P}_{22} \end{bmatrix} \begin{bmatrix} X \\ U \end{bmatrix} \\
= -\hat{R}^{-1} \hat{P}_{12}^T X - \hat{R}^{-1} \hat{P}_{22} U$$

LQR Extensions: 3. Weightage on Rate of Control

However, $\dot{U} = -\hat{R}^{-1}\hat{P}_{12}^{T}X - \hat{R}^{-1}\hat{P}_{22}U$ is a dynamic equation in U and hence is not easy for implementation. For this reason, we want an expression in the RHS only as a function of X and operations on it.

State equation: $\dot{X} = AX + BU$

This suggests: $U = B^+(\dot{X} - AX)$

(Note: This is only an approximate solution, unless $m \ge n$)

LQR Extensions: 3. Weightage on Rate of Control

$$\dot{U} = -\hat{R}^{-1}\hat{P}_{12}^{T}X - \hat{R}^{-1}\hat{P}_{22}B^{+}\dot{X} - \hat{R}^{-1}\hat{P}_{22}B^{+}A\dot{X}$$

$$= -\hat{R}^{-1}(\hat{P}_{12}^{T} + \hat{P}_{22}B^{+}A)X - \hat{R}^{-1}\hat{P}_{22}B^{+}\dot{X}$$

$$= -K_{1}\dot{X} - K_{2}X$$

Integrating this expression both sides,

$$U = -\underbrace{K_1 X}_{\text{Proportional}} - K_2 \int_{0}^{t} X(z) dz + \underbrace{U_0}_{\text{Initial condition}}$$

Note: U_0 can be obtained using a performance index

without the \dot{U} term

4. Prescribed Degree of Stability

Condition: All the Eiganvalues of the closed loop system should lie to the left of line *AB*

$$J = \frac{1}{2} \int_{t_0}^{\infty} e^{2\alpha t} \left[X^T Q X + U^T R U \right] dt \quad \text{where, } \alpha \ge 0$$

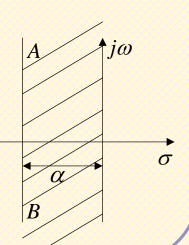
$$=\frac{1}{2}\int_{t_0}^{\infty} \left(\left[e^{\alpha t} X \right]^T Q \left[e^{\alpha t} X \right] + \left[e^{\alpha t} U \right]^T R \left[e^{\alpha t} U \right] \right) dt$$

$$=\frac{1}{2}\int_{t_0}^{\infty} \left(\tilde{X}^T Q \tilde{X} + \tilde{U}^T R \tilde{U}\right) dt$$

Let
$$\tilde{X} = e^{\alpha t} X$$

 $\tilde{U} = e^{\alpha t} U$

Co-ordinate transformation



4. Prescribed Degree of Stability

$$\tilde{X} = e^{\alpha t} \dot{X} + \alpha e^{\alpha t} X$$

$$= e^{\alpha t} (AX + BU) + \alpha e^{\alpha t} X$$

$$= A(e^{\alpha t} X) + B(e^{\alpha t} U) + \alpha (e^{\alpha t} X)$$

$$\dot{\tilde{X}} = (A + \alpha I) \tilde{X} + B\tilde{U}$$
Control Solution: $\tilde{U} = -K\tilde{X}$

$$e^{\alpha t} U = -Ke^{\alpha t} X$$

$$U = -KX$$

4. Prescribed Degree of Stability

Modified System:

$$|\tilde{U} = -K\tilde{X}|$$

$$\left|\dot{\tilde{X}} = \left[\left(A - BK \right) + \alpha I \right] \tilde{X}\right|$$

Actual System:

$$U = -KX$$

$$\dot{X} = (A - BK)X$$

K is designed in such a way that eigenvalues of

$$\lceil (A - BK) + \alpha I \rceil$$
 will lie in the left-half plane.

Hence, eigenvalues of (A - BK) will lie to the left of a line parallel to the imaginary axis, which is located away by distance α from the imaginary axis.

LQR Design for Command Tracking

Dr. Radhakant Padhi

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Dept. of Aerospace Engineering

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LQR Design for Command Tracking

Problem:

To design U such that a part of the state vector of the linear system $\dot{X} = AX + BU$ tracks a commanded reference signal.

i.e.
$$X_T \to r_c$$
, where $X = \left\lceil \frac{X_T}{X_N} \right\rceil$

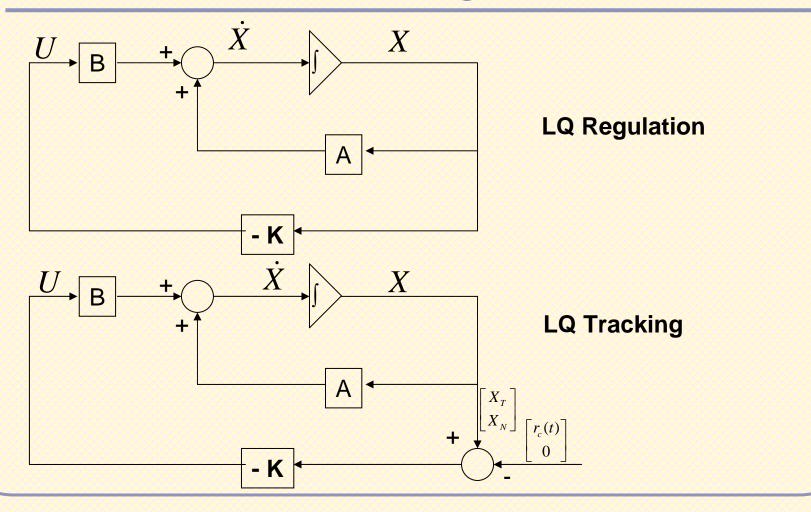
Solution:

1) Formulate a standard LQR problem. Typically Q However, select the Q matrix properly.

Typically
$$Q = \begin{bmatrix} Q_{TT} & 0 \\ 0 & 0 \end{bmatrix}$$

2) Implement the controller as
$$U = -K \begin{bmatrix} X_T - r_c \\ X_N \end{bmatrix}$$

LQR Design for Command Tracking



LQR Design for Command Tracking with Integral Feedback

Solution (with integral controller):

1) Augment the system dynamics with integral states

$$\begin{bmatrix} \dot{X}_T \\ \dot{X}_N \\ \dot{X}_I \end{bmatrix} = \begin{bmatrix} A_{TT} & A_{TN} & 0 \\ A_{NT} & A_{NN} & 0 \\ I & 0 & 0 \end{bmatrix} \begin{bmatrix} X_T \\ X_N \\ X_I \end{bmatrix} + \begin{bmatrix} B_T \\ B_N \\ 0 \end{bmatrix} U$$

- 2) Select the Q matrix properly (should penalize only X_T and X_I states)
- 3) Control solution $U = -K \left[(X_T r_c)^T \quad X_N^T \quad \left(\int_0^t (X_T r_c) dt \right)^T \right]^T$

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Reference

Optimum Intercept Laws for Accelerating Targets

VITALIJ GARBER*

AIAA JOURNAL VOL. 6, NO. 11
NOVEMBER 1968

• To derive the state X of a linear (rather linearized) system $\dot{X} = AX + BU + C$ to the origin by minimizing the following quadratic performance index (cost function)

$$J = \frac{1}{2} \left(X_f^T S_f X_f \right) + \frac{1}{2} \int_{t_0}^{t_f} \left(X^T Q X + U^T R U \right) dt$$

where

$$S_f, Q \ge 0$$
 (psdf), $R > 0$ (pdf)

Performance Index (to minimize):

$$J = \frac{1}{2} \left(X_f^T S_f X_f \right) + \frac{1}{2} \int_{t_0}^{t_f} \left(X^T Q X + U^T R U \right) dt$$

• Path Constraint: $\dot{X} = AX + BU + C$

• Boundary Conditions: $X(0) = X_0$: Specified t_f : Fixed, $X(t_f)$: Free

- Terminal penalty: $\varphi(X_f) = \frac{1}{2} (X_f^T S_f X_f)$
- Hamiltonian: $H = \frac{1}{2} (X^T Q X + U^T R U) + \lambda^T (AX + BU + C)$
- State Equation: $\dot{X} = AX + BU + C$
- Costate Equation: $\dot{\lambda} = -(\partial H / \partial X) = -(QX + A^T \lambda)$
- Optimal Control Eq.: $(\partial H / \partial U) = 0 \implies U = -R^{-1}B^T \lambda$
- Boundary Condition: $\lambda_f = (\partial \varphi / \partial X_f) = S_f X_f$

Riccati equation

$$\dot{P} + PA + A^T P - PBR^{-1}B^T P + Q = 0$$

Auxiliary equation

$$\left| \dot{K} + \left(A^T - PBR^{-1}B^T \right) K + PC = 0 \right|$$

Boundary conditions

$$P(t_f)X_f + K(t_f) = S_f X_f \quad (X_f \text{ is free})$$

$$P(t_f) = S_f \quad K(t_f) = 0$$

Control Solution:

$$U = -R^{-1}B^{T}\lambda$$

$$= -R^{-1}B^{T}(PX + K)$$

$$= -R^{-1}B^{T}PX - R^{-1}B^{T}K$$

Note: There is a residual controller even after $X \to 0$. This part of the controller offsets the continuous disturbance.

